PROJECT DOCUMENT COVER SHEET

INFLIGHT ASSESSMENT OF CARDIOVASCULAR FUNCTION

M051

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DB-53-MO51

DATE

April 28, 1967

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EXPERIMENT IMPLEMENTATION PIAN FOR MANNED SPACE FLIGHT EXPERIENT

1.	Title of Experiment	Experiment No. M-051
TI	NFLIGHT ASSESSMENT OF CARDIOVASCULAR	FUNCTION
2.	Sponsorship	
	Sponsoring Program Office (SPO)	OMSF, Space Medicine
	Principal Investigator(s) (PI)	Approved
		Date
	PI Institutional Affiliation	NASA, MSC Biomedical Research Office, DB
	SPO Manager	S. P. Vinograd, M.D.
3.	Development	
	Experiment Development Center	Approved
		Robert F. Thompson, MSC Date
	Director of Medical Research	5400
	and Operations	Approved
		Charles A Berry, M.D.
	Center Experiments Manager Chief, Biomedical Research Office	Approved . 7. Dietlein, M.D.
	onicity promote the contract of	Date
	Integration	
	OMSF Flight Program Office	ML
	Payload Integration Center	Approved
		Leland F. Belew, MFSC
		D.10C
	Approval (Headquarters use only)	
	Sponsoring Program Office Approval_	Date
	Flight Program Office Approval	Date
	Approved for Flight Assignment	Date
	Ass	Program Office sociate Administrator

Date

SECTION I - EXPERIMENT SUMMARY

- 1.0 TECHNICAL INFORMATION:
- 1.1 Purpose and Objective:
- 1.1.1 To provide evaluation of cardiovascular and hematologic systems pre- and postflight which will quantitate the changes in these systems resulting from prolonged space flight.
- 1.1.2 To provide information concerning the time course of cardio-vascular deconditioning during flight and inflight data for predicting the degree of orthostatic intolerance and impariment of physical capacity to be expected following return to Earth's environment. The accomplishment of this objective is predicated upon the development and validation of the lower body negative pressure (LBNP) equipment (weightlessness tilt table analog) and the presence of a physician crewmember.

The detection of space flight cardiovascular deconditioning (degradations in cardiovascular function which may impair its performance during space flight or after return to Earth's environment) and the establishment of the time course of these changes constitute the purpose of this experiment.

Crewmen of Projects Mercury and Gemini, as well as subjects of recumbency and immersion studies, have exhibited a reduced orthostatic tolerance when exposed to the stress of passive tilt table studies during the postflight, postrecumbency, or postimmersion period. Characteristic features of this orthostatic response are cardioacceleration, lower extremity pooling of blood, and decreased pulse pressure. Severe orthostatism invariably results in syncope (loss of consciousness). Additionally, loss of body water (evidenced by weight loss), decrease in red blood cell mass, decrease in red blood cell survival time, and increase in red blood cell fragility have been demonstrated postflight in Gemini flight crews.

This investigation will provide insight into the following specific questions: Is there progressive cardiovascular deterioration in a weightless environment? If so, is it uniform or sporadic? At what level must it be checked in order to assure tolerance of reentry by the astronaut? How may such deterioration be checked? Can a preconditioning routine be developed to minimize inflight effects?

2.0 ENGINEERING INFORMATION:

2.1 Major Hardware Components:

- 2.1.1 VCG electrodes and harness (vest); present Apollo system or quick donning type, one set per astronaut with onboard readout of heart rate.
- 2.1.2 Signal conditioners, present Apollo type, one set.
- 2.1.3 Blood pressure cuff, improved Gemini type with onboard readout, one set.
- 2.1.4 Lower body negative pressure device (LBNP), in development, one set.
- 2.1.5 Vacuum source with suitable controls, safety devices, valves, and gauges for evaluation of LBNP.
- 2.1.6 Cage capacitance plethysmograph, or modified Mercury strain gauge, for measurement of leg volume, two units.
- 2.1.7 Ear canal temperature probe, stowage box 20" x 24" x 12".
- 2.1.8 Bicycle ergometer (from M050).
- 2.1.9 Experiment Data Acquisition System (EDS): Full data recording and dumping capability for biomedical data from MO18, MO50, MO51, MO53, M486, M488, M489 will be provided by either the OWS data system or by a separate piece of operating equipment.

2.2 Characteristics of Major Components:

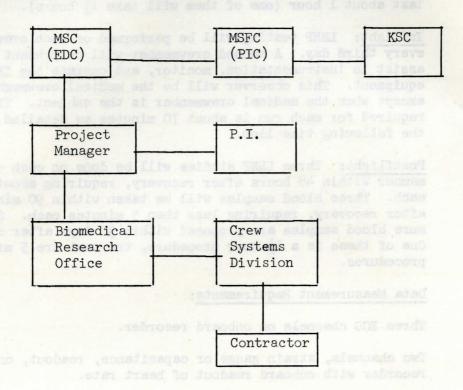
		Size	Vol.C.F.	Wt. #	Power
(1)	Vest & Signal Conditioners (3)	12"x12"x13"	0.9	5.1	*
(2)	Blood Pressure Cuff	6" cyl, 6" lg	0.1	0.5	* *
(3)	LBNP (with vac. hose) Gauges	20"x15"x12" 3" dia x 2" le		20.0	*
(5) (6)	Stowage box EDS (available from	20"x24"x12"	(3.34)	10.0	Rec. 10W
	onboard operating equipment)				TM 2W
(7)	-	3/4"x3/8"x3/8"		.3	
	Total:		3.3	37.9	12W

^{*}Power for data handling included in EDS

- 3.0 OPERATIONS SUPPORT:
- Preflight: Blood and LBNP studies will be done three times, at weekly intervals prior to flight. These sessions will last about 1 hour (one of them will take $1\frac{1}{2}$ hours).
- Inflight: LBNP testing will be performed on each crewmember every third day. A second crewmember will be present to assist in instrumentation, monitor, and operate the LBNP equipment. This observer will be the medical crewmember except when the medical crewmember is the subject. Time required for each run is about 70 minutes as detailed on the following time line.
- Postflight: Three LBNP studies will be done on each crewmember within 48 hours after recovery, requiring about 1 hour each. Three blood samples will be taken within 90 minutes after recovery, requiring less than 5 minutes each. Seven more blood samples are proposed within 30 days after recovery. One of these is a l_2 hour procedure, the rest are 5 minute procedures.
- 3.4 Data Measurement Requirements:
- 3.4.1 Three ECG channels on onboard recorder.
- 3.4.2 Two channels, strain gauge or capacitance, readout, onboard recorder with onboard readout of heart rate.
- 3.4.3 One voice annotation channel on onboard recorder.
- 3.4.4 One time hack channel, onboard recorder.
- 3.4.5 Onboard readout of LBNP negative pressure gauge.
- 3.4.6 Written entries in onboard log.
- 3.4.7 Onboard readout of blood pressure gauge.
- 3.4.8 Onboard readout of leg volume changes.
- 3.4.9 Onboard readout of ear canal temperature.

Data collection will be accomplished by both onboard tape recording with fast dump capabilities and onboard readout to permit both ground-based monitoring of flight crew status and onboard evaluation by the physician crewmember.

4.0 MANAGEMENT ARRANGEMENTS: The major organizational elements involved in implementation of the experiment and their relationship with the Principal Investigator, the equipment contractor, and the space vehicle contractor are as follows:



5.0

SCHEDULE AND RESOURCE REQUIREMENTS:

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SECTION II - TECHNICAL APPROACH

1.0 OBJECTIVES:

- 1.1 To provide evaluation of cardiovascular and hematologic systems pre- and postflight which will quantitate the changes in these systems resulting from prolonged space flight.
- 1.2 To provide information concerning the time course of cardio-vascular deconditioning during flight and provide inflight data for predicting the degree of orthostatic intolerance and impariment of physical capacity to be expected following return to Earth's environment. The accomplishment of this objective is predicated upon the development and validation of the lower body negative pressure (LBNP) equipment (weightlessness tilt table analog) and the presence of a physician crewmember.

2.0

JUSTIFICATION: The detection of space flight cardiovascular deconditioning (degradations in cardiovascular function which may impair its performance during space flight or after return to Earth's environment) and the establishment of the time course of these changes constitute the purpose of this experiment.

Crewmen of Projects Mercury and Gemini, as well as subjects of recumbency and immersion studies, have exhibited a reduced orthostatic tolerance when exposed to the stress of passive tilt table studies during the postflight, postrecumbency, or postimmersion period. Characteristic features of this orthostatic response are cardioacceleration, lower extremity pooling of blood, and decreased pulse pressure. Severe orthostatism invariably results in syncope (loss of consciousness). Additionally, loss of body water (evidenced by weight loss), decrease in red blood cell mass, decrease in red blood cell survival time, and increase in red blood cell fragility have been demonstrated postflight in Gemini flight crews.

3.0 EXPERIMENT APPROACH:

Procedure: Generally, it is anticipated that three preflight control data acquisition periods (T-21, T-10, and T-2) will be required. During these sessions, blood studies will be accomplished. These procedures are in essence the same as those performed during Project Gemini and are again essentially the same as those required for Project Apollo Experiment MOlla and MO22.

The exercise testing and the baseline electrolytes will be accomplished within the context of the "Metabolic Cost of Inflight Tasks" (MO50) and "Bone and Muscle Changes" (MO52) experiments.

The postflight data acquisition periods should be accomplished as soon as possible, 24 hours and 48 hours after splashdown. The procedures will in essence be identical to the preflight periods. The LBNP procedure should be performed during the morning hours and each subject should have all LBNP tests at similar times of the test day. This test should be performed fasting or at least 3 hours after eating. It should not be preceded by vigorous exercise nor should it be performed while unduly fatigued or in high environmental temperatures. Inflight measurements during LBNP forcing will be made only if a physician is one of the crewmembers. Measurements should be obtained on each man every third day.

Lower body negative pressure procedures is as follows: After the proper bioinstrumentation has been placed on the subject, he will insert the legs and hips inside the LBNP device. The waist seal will then be adjusted and secured. Whether the cage capacitance plethysmographs can be placed in position most conveniently before or after getting into the LBNP device will be determined. The subject should be in such a position that he can rest comfortably without motion in an extended stance.

The baseline of each of the recording channels should be adjusted to the proper position on the strip chart and calibration signals recorded on all channels.

The procedure will consist of three phases. The first phase is a 5-minute control phase; the second, a 15-minute negative pressure phase; and the third, a 5-minute supine recovery phase. VCG, ear canal temperature, and leg plethysmograph recordings will be made continuously throughout the procedure. Blood pressure will be measured every 30 seconds. To insure a completely passive subject he should be requested to remain immobile during the 25 minutes that recordings are being made.

Parameters to be Measured: The broad experimental plan is depicted by Table 1. The major stress imposed upon the system under analysis is the space mission itself; the known load stresses imposed upon the system are exercise and lower body negative pressure.

TABLE 1

Preflight	Inflight	Postflight
Assessment of Baseline System Performance (Controls)	Time Course Assess- ment of Changes Caused by Major Stress	Time Course Assessments of Return to Con- trol Values
a. Basic System Parameters b. Responses to Load Stresses	a. Basic Parame- tersb. Responses to Load Stresses	a. Basic System Parameters b. Responses to Load Stresses

Table 2 lists the parameters to be measured. The arrangement indicates the broad time relationship, the basic parameters for general system performance assessment and the specific parameters to be measured during stress loads. Expected values are indicated at the appropriate intersections of condition and parameter. Qualitative or interpretive parameters, for example, ECG and thoracic X-ray, will be evaluated by Normal (N) or Change (C).

TABLE 2

a. Preflight and Postflight Assessment of System Performance

	Rest	Load Stres Exercise	ses LBNP
D			
Basic Parameters Thoracic X-ray	erred N		
Blood Volume	3.5-5.1		
RBC Mass	1500-2200 ml.		
Plasma Volume	2000-2800 ml.		
Electrolytes	N		
Forcing Parameters			
Electrocardiogram	N	N	N
Heart Rate	40-90		
Skin Temperature	87°-95° F.	Action the Later	//
Blood Pressure	100/70	200/110	90/40
Leg Volume, % Chang	е		4-5
LBNP			5 60 -4

b. Inflight Assessment of System Performance

aminaemini etriaman	Dogt	Load Stres	ses	
ovideraneda) to evident.	Rest	Exercise	LBNP	
Basic Parameters				
Fluid Intake/Output	N			
Electrolytes	C			
Forcing Parameters				
Electrocardiogram	N	N	N	
Heart Rate	40-90	70-180	70-180	
Skin Temperature	60°-99° F.	700/50	00/55	
Blood Pressure	100/60 -	100/70 - 180/90	90/55 - 180/100	
	120/70	100/90	100/100	

Research programs have established the required pressure gradient (LBNP) to simulate lg tilt and the rates of "deconditioning" which might be expected during 30 days of weightlessness as simulated by bed rest or shorter periods of water immersion.

A research program would validate the experimental proceedings utilized for parameter measurements and attempt to isolate the effects of weightlessness from those of atmosphere differences. 4.0

DISCIPLINARY RELATIONSHIPS: Since the application of LBNP is not gravity dependent, it is possible to employ it in space flight as an "orthostatic" test to reveal the time course and magnitude of changes in circulatory regulation known to occur during adaptation to space flight.

Heart rate (HR) and electrocardiographic (VCG) data are to be acquired by means of slightly modified Apollo instrumentation. The major modification requirements are the incorporation of a means of calibrating the ECG (calibration to be applied across the input electrodes), a quick don, quick doff electrode and harness system, and a means of identifying the crewmemberf from which data is being collected.

Blood pressure data acquisition will require improvement. An acceptable technique which yields both onboard readout and a recordable output will be developed.

The electrolyte and fluid intake/output data for this study will be obtained from the experiment entitled "Bone and Muscle Changes During Prolonged Space Flight." No additional requirements are imposed.

The recording of heart rate, skin temperature, and the external workload constitute the requirement for evaluating the responses of the cardiovascular system to exercise. These requirements will be satisfied by the equipment involved in the experiment entitled "Metabolic Costs of Inflight Tasks."

The remaining studies are conducted pre- and postflight. These are standard Earth-based laboratory studies for which methodologies and instrumentation are developed and are in general use.

5.0

BASELINE OR CONTROL DATA: Laboratory equipment for the LBNP experiment is currently available and in use at USAF SAM, at Douglas Aircraft, Inc., and at Wilfort Hall USAF Hospital. Pre- and postflight uses of LBNP have been proposed by USAF as a Block I Apollo Experiment (MO23). Baseline data proving the validity of the experiment are currently being collected at the Wilford Hall USAF Hospital. This background experimentation will subject test subjects to stresses similar to those to be encountered by the astronauts. Bed rest will be used to simulate weightlessness. Data collected to date indicate that LBNP results in the pooling of blood and shifts in position of viscera that are similar to the effects of being passively positioned with the long axis of the body parallel to the Earth's gravitational field (as in a tilt table test). The circulatory adjustments resulting from LBNP are strikingly similar to those resulting from tilt procedures. Furthermore, at a given pressure gradient, the response is more reproducible than the tilt response and is less subject to artifact. Serial steps of increasing differential pressure may be applied, with reproducible evidence of increasing circulatory stress. The tilt table procedure does not allow this flexibility in adjusting the degree of "orthostatic" stress.

SECTION III - ENGINEERING INFORMATION

- 1.0 EQUIPMENT DESCRIPTION:
- 1.1 Heart Rate and VCG:
- 1.1.1 Sensors (Electrodes) and Harness: Apollo sensors can be used. Development of quick don and doff sensor and harness system would increase the operational efficiency of the experiment.
- 1.1.2 Signal Conditioners: Flight units are being developed by Spacelabs, Inc., and are scheduled for delivery by November 1967. A means of calibration and crewmember identification will be provided.
- Blood Pressure Cuff: Modification of the present Gemini system to provide increased reliability and an onboard readout will be necessary.
- Lower Body Negative Pressure Device (LBNP): A laboratory model has been developed by Lockheed for the USAF. Douglas Aircraft also has a laboratory model. The USAF is developing flight type equipment. A contract is being negotiated.

The device consists of a bag surrounding a framework which may be collapsed for storage, a sealing membrane to fit about the midsection of a man, an evacuation device (to create the required pressure gradient) with suitable controls, valves, and safety devices. A pressure sensor is incorporated to provide recording of tape and an onboard readout of the negative pressure.

- Cage Capacitance Plethysmographs or Modified Mercury Strain
 Gauges: Two units, one for each leg, are required.
- 1.5 Ear Canal Temperature Probe.
- 1.6 Stowage Box Approximately 20" x 24" x 12".
- Medical Data Acquisition System (Separate onboard operating equipment not a part of this experiment): A medical data acquisition system will be needed for this and other medical experiments proposed for this flight. The Gemini pulse code modulation system will satisfy part of the requirements for the medical data acquisition system but the frequency response (8 cps) is not adequate for ECG, EEG, and voice. Thus, an additional recorder and telemetry system is needed. For correlation purposes, mission elapsed time, and subject identity would also be necessary on the second recorder.

1.8 Characteristics of Major Components:

3	tion units. Quali	Size	Vol. Ft.3	Wt.#	Power
1.	Vest & Signal	12"x12"x3"	0.9	5.1	*
2	Conditioners (3) Blood Pressure	15 X15 X2	0.9	d With I	
	Cuff	6"cyl, 6" lg	0.1	0.5	*
3.	LBNP (with vac. hose)	20"x15"x12"	2.1	20.0	*
4.	Gauges	3" dia x 2" lg	0.2	2.0	*
6.	Ear Canal Temperature Probe Stowage Box	3/4"x3/8"x3/8" 20"x24"x12"	(3.34)	0.3	
7.	EDS (available from onboard operating equipme	ent)			Rec. 100V
	Total	:	3.3	37.9	121

*Power for data handling included in EDS

1.9 Quantities Required:

	Vest & Signal Conditioners	BP Cuff	LBNP Vac. Source	EC Temp. Probe	Gauge Set
Mass Mockup Training Qualification Flight Backup	3 3 1 6	1 1 2 1	1 1 1 1	3 3* 1 3*	1 1 1 1

*Ear Canal Temperature Probe will be molded to fit each individual crewmember.

1.10 Special GSE or Logistic Support: None.

1.11 Definition:

- 1.11.1 Mass Mockup: This unit is an exact model with correct external dimensional configuration and with weight, center-of-gravity, functional control, electrical connections, and mounting provisions of the flight hardware. This need not be a functional model.
- 1.11.2 Training Hardware: These units are of the flight configuration to provide realistic operation and manipulation characteristics and to determine compatibility with other crew equipment and activities. Training units must have operational characteristics identical to flight equipment.

- 1.11.3 Qualification Test Hardware: This is a flight qualifiable unit selected from the first production units. Qualification testing is performed on this unit.
- 1.11.4 Flight and Backup Hardware: These units are flight items and must be identical in design, configuration, and production processing to the finally accepted qualification test unit. Any changes to the qualification test unit during qualification testing must be included in the flight and backup units prior to delivery to NASA MSC. These units must be manufactured using the same production techniques as the qualification test unit.

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Mass Mosson: This and the an exact model with correct

of-gravity, functional control, electrical connections, and mounting provisions of the filight hardware. This need not

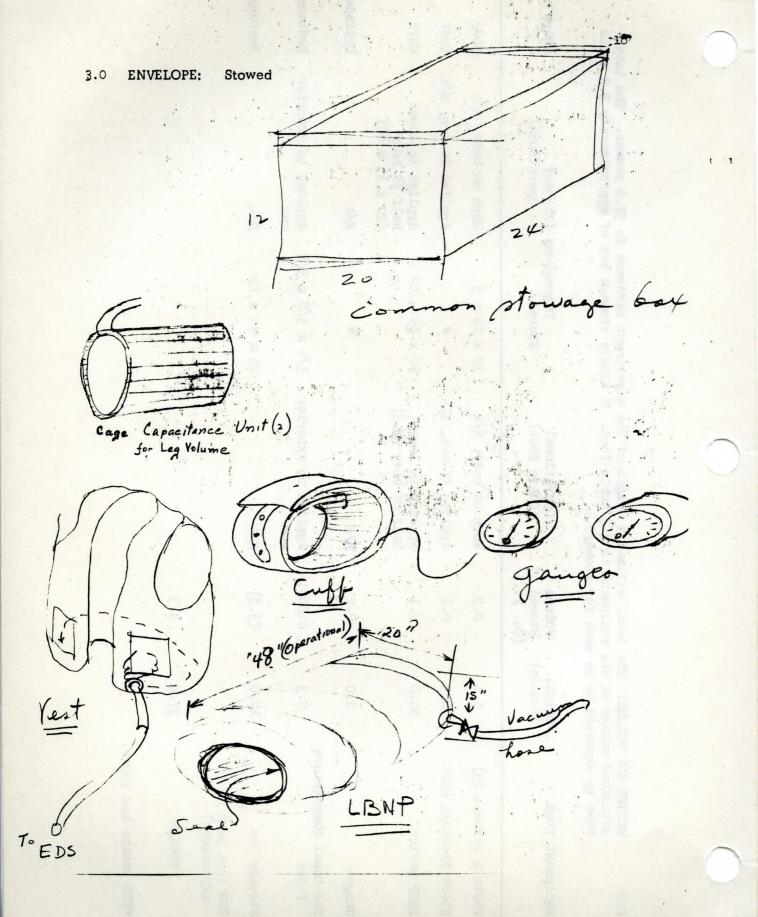
Training Hardware: These units are of the flight configura-

er ev squignost and soblytules. Training mile sea baye

WEIGHT AND VOLUME: The total estimated weight of the experiment hardware is 37.9 pounds. The total estimated volume of the equipment is 3.3 cubic feet. A single stowage box of approximately 3.3 cubic feet is expected to be used for stowage. 5.0

Shape	Body	A Cuff	Cuff	Cylinder	r Rectangle	Rectangle		
(Inches) Operational	Worn on body, N/A	Applied to arm, N/A	Applied to lower half of body (20 x 15 x 48)	N/A	Applied to exterior ear	N/A		
Dimensions (Inches) Stored Opera	12 x 12 x 3	*	20 x 15 x 12	*	3/4 × 3/8 × 3/8	20 x 24 x 12		
Operational (Cubic Feet)	Worn on body, N/A	Applied to arm, N/A	Applied to lower half of body (3.3)	N/A	Applied to exterior ear	N/A		
Volume Stored (Cu. Ft.)	6.0	0.1	2•1	0.2	< 0.0001	(3.3)		3.3
Weight (Pounds)	5.1	0.5	20.0	2.0	0.3	10.0		37.9
Equipment Item	Vests & Harness (3)	Blood Pressure Cuff	LBNP Device	Gauges	Ear Canal Temperature Probe	Stowage Box	EDS (not part of experiment)	TOTALS

*Fit common box shown.



- 4.0 SPACECRAFT INTERFACE REQUIREMENTS:
- 4.1 Stowage for launch should be in the Multiple Docking Adapter (MDA). In any case, EVA should not be necessary to retrieve equipment. The universal racks planned for the MDA and OWS will provide adequate stowage mounting.
- 4.2 Electrical power will not be required except for data system shown below. Data monitoring and connection power will come from separate data system.
- 4.3 Plumbing modifications are required for vacuum line to LBNP.
- 4.4 Life support services are not required.
- 4.5 Biomedical and systems data will be provided through the umbilical to the Airlock Module.

FOWER: Recorder 10W
TM 2W

Stowage for launch should be in the Multiple Docking Aless (MCA). In any case, EVA should not be decessary to retrive equipment. The universal rocks planned for the MDA and U

Electrical power will not be required except for data system whom below. Data most boring and commented power will come

Physhing modifications are required for varuum line to LEGP.

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6.0 ENVIRONMENTAL CONSTRAINTS:

Constraint	Range	
Thermal ° C.		
Stored	-51 to +70	
Operational	-51 to +70	
Tolerance Levels	5000	-6
Pressure	25 psia to 1 x 10	mm Hg
Relative Humidity	15% to 100%	
Atmospheric Composition Contaminants	20% to 100% 0 ₂	
Acceleration (Storage) Positive Negative Transverse	O to 7g	
Acceleration (Operation)	0 to 0 la	
Positive	0 00 0.18	
Negative		
Transverse		
Vibration (Storage)		
Random	20 to 100 cps	linear increase (log by log_plot) from 0.00lg /cps to 0.075g ² /cps
	100 to 500 cps	constant at 0.075g ² /cps
	500 to 1000 cps	linear decrease (log by log plot) from 0.075g ² /cps
	1000 to 2000 cps	to 0.015g ² /cps Constant at 0.015g ² /cps
Shock	30g	

7.0 DATA MEASUREMENT REQUIREMENTS: The following table presents data handling requirements

1000 to 2000 cps

Test Time Span	Parameter	Band Width (Hz)	Range	Accuracy
25 min.	VCG	0-100	O-5VDC	±3%
	VCG	0-100	O-5VDC	±3%
	VCG Blood	0-100	0-5VDC	±3%
	Pressure	0-50	O-20MC	±3%
			differential	
	LBNP	0-0.25	O-5VDC	±3%
	Leg Volume	0-0.25	O-5VDC	±3%
	Leg Volume	0-0.25	O-5VDO	±3%
	Body Temp.	0-0.25	O-5VDC	±3%
	Timing Voice			±0.1 sec.

SECTION IV - RELIABILITY AND QUALIFICATION REQUIREMENTS

The experiment equipment and associated hardware will be controlled and measured against acceptable standards of workmanship and design to assure system operational integrity and minimum degradation of experiment objectives. Such standards and measurements will be verified by implementing in-process inspection techniques and Government monitoring of qualification tests, including certification histories, as required.

Preparation of the final R and QA program will be closely coordinated with the payload integration center to insure adequate provisions for flight safety and experiment success.

SECTION V - OPERATIONS SUPPORT

1.0 SPACEORAFT ORIENTATION REQUIREMENTS: None.

2.0 ASTRONAUT TRAINING: Each astronaut will be trained in procedures and techniques while baseline control data is being obtained. Total training required is as follows:

	During Baseline Testing	Other Training	Total Per Man
Astronaut 1	3 Hrs.	2 Hrs.	5 Hrs.
Astronaut 2	3 Hrs.	2 Hrs.	5 Hrs.
Physician/Astronaut	3 Hrs.	10 Hrs.	13 Hrs.

3.0	ASTRONAUT TRAINING SYNOPSIS:
3.1	Training in proper application of electrodes .
3.2	Donning and fastening vest.
3.3	Operation of signal conditioners.
3.4	Method of calibrating equipment.
3.5	EDS interfaces.
3.6	Application of capacitance plethysmograph.
3.7	Application of ear canal temperature probe.
3.8	Application of LBNP device and body seals.
3.9	Watching BP and HR responses to gain experience in evaluating responses.
3.10	Timeline table follows:

TIMELINE TABLE

Subject	Removes outer garments	Extends Lower Body Negative Pressure Device and secures to working area.	Removes 8 electrodes from Electrode Application Kit and connects each electrode to appropriate outlets in Vest.			Prepares skin sites for electrode application over anterior chest.	saunaid bee mov peliaulba et elaise d	Dons Vest and Harness.	Assists in electione applications
Time (Min.) Cum. Time Unit Time	0:00/5:00	5:00/0:30	5:30/4:00			9:30/2:00		11:30/0:30	14:00/4:00
Observer	Obtains from storage compartment: a. Blood Pressure Apparatus b. Harness and Vest c. Electrode Application Kit d. Ear Canal Temperature Probe e. Biomedical Cable f. 2 Cage Capacitance Plethyomographs	Connects Biomedical Cable to EDS.	Turns knobs and switches on EDS to activate Tape Recorder, Display System, and Vacuum Source.	Assembles units of Temperature Recording Device.	Assembles units of Blood Pressure Apparatus.	Prepares skin sites for electrode application on back, one leg, and axillae of subject.	Applies electrodes to skin		
Time (Min.) Cum.Time/	0:00/2:00	5:00/0:30	5:30/0:30	6:00/1:00	7:00/1:00	8:00/3:00	11:00/3:00		

Adjusts and tightens Vest and Harness. Connects Biomedical Cable to Conditioners in Vest and actional Conditioners. Checks each VCG channel to tain proper function by transito display system. Applies Plethysmograph Unitleg. Applies Plethysmograph Unitleg. Calibrates each Plethysmograssecuring gages with adhesive Observes display to calibrate and connects to Biomedical Gand Connects output from Blood Fdevice to Biomedical Cable.	Time (Min.) Cum.Time/ Unit Time		o Signal 14:30/0:30 Assembles Strain Gage Unit for left leg.	ascer- 15:00/0:30 Assembles Strain Gage Unit for right leg.	15:30/0:30 Steps into LBNP enclosure and assumes comfortable position with legs extended and waist seal loose about waist.	to left 16:00/2:00 Stands or reclines quietly.	to right 18:00/2:00 Connects outlet from left Strain Gage to Biomedical Cable.	ph after 20:00/1:00 Connects outlet from right Strain e. Gage to Biomedical Cable.	21:00/1:00 Avoids leg movement during and after calibration.	device $22:00/1:00$ Applies Blood Pressure Apparatus to $^{\infty}_{\circ}$ Jable.	ressure 23:00/3:00 Remains motionless except for upper extremities while assisting in securing
	Ohserver	Adjusts and tightens Vest and Harness.	Connects Biomedical Cable to Signal Conditioners in Vest and activates Signal Conditioners.	Checks each VCG channel to ascertain proper function by transmitting to display system.		Applies Plethysmograph Unit to left	Applies Plethysmograph Unit to right leg.	Calibrates each Plethysmograph after securing gages with adhesive. Observes display to calibrate.		Attached temperature sensing device and connects to Biomedical Cable.	Connects output from Blood Pressure device to Biomedical Cable.

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1	1	4

Time (Min.) Cum. Time/ Unit Time		Time (Min.) Cum. Time/ Unit Time	
23:30/1:30	Checks signals from Temperature Sensing Device and Blood Pressure Apparatus to ascertain proper function.		
25:00/3:00	Secures waist seal.		
28:00/1:00	Sets LBNP regulator to -10mm Hg, activates vacuum source, and checks seal and pressure gage. Releases vacuum.	E3:00/2:00	Samones Nest and Horness, Cleaness
29:00/2:00	Checks each channel and records calibration signals for each.		
31:00/5:00	Records all channels continuously throughout 5 minutes.	31:00/5:00	Remains motionless during 5 minutes of control data acquisition.
36:00/0:30	Calibrates all channels (1 button). Sets vacuum regulator to -40 (?50) mm Hg and opens vacuum system to evaluate LBNP enclosure to -40 mm Hg pressure.	36:00	Compresses safety release device and maintains this compression throughout application of negative pressure. Remains motionless during 15 minutes.
36:30/15:00	Observes Blood Pressure (2x/min.), heart rate (continuously with 10 second means) on display system and appearance of subject. Comments as indicated into voice channel. Calibrates system every 2 minutes.	83/190\7:00	Device from ear.
51:30/5:00	Open LBNP system and closes vacuum source. Continues recording for 5 minutes after release of LBNP. Calibrates at 2 and 4 minutes.	51:30/5:00	Releases safety device. Remains motionless during 5 minutes recovery period.

Time (Min.) Cum.Time/	Carreston & Good Wings & Control of the Lands of the Land	Time (Min.) Cum.Time/ Unit Time	Subject
56:30/1:00	Turns off recording system and dis- connects Biomedical Cable at EDS	56:30/1:00	Loosens waist seal and Blood Pressure Apparatus.
57:30/1:00	Completes removal of waste seal. Disconnects Blood Pressure output from Cable and removes apparatus from subject's arm. Stows in container.	57:30/1:00	Disconnects Temperature Sensing Device from ear.
58:30/1:30	Removes both Plethysmograph elements from subject's legs. Disconnects from Biomedical Cable. Stows in container.	58:30/1:30	Begins removal of chest electrodes.
60:00/1:00	Disconnects Temperature Sensing Device from Biomedical Cable. Removes and stows in container.	60:00/1:00	Steps out of LBNP enclosure. Continues removal of chest electrodes.
61:00/1:00	Removes remaining electrodes from back, leg, and axillae.	61:00/2:00	Cleans electrodes and returns them to Electrode Application Kit.
62:00/1:00	Returns LBNP device to folded-up mode and secures.	63:00/2:00	Removes Vest and Harness. Cleanses skin sites and discards disposable material in waste container.
63:00/3:00	Stows Cable, Electrode Application Kit, and Vest-Harness in container and places in storage.	65:00/1:00	Dons outer garments.
66 minutes		66 minutes	

3.11 Summary of Astronaut Activities:

	Preflight	Inf	Postflight	
		Subject	Observer	
Astronaut #1 Astronaut #2 Physician/Astronaut	5 5 13	4.8 4.8 4.8	4.8 4.8 4.8	3 3 3
	23	14.4	14.4	9

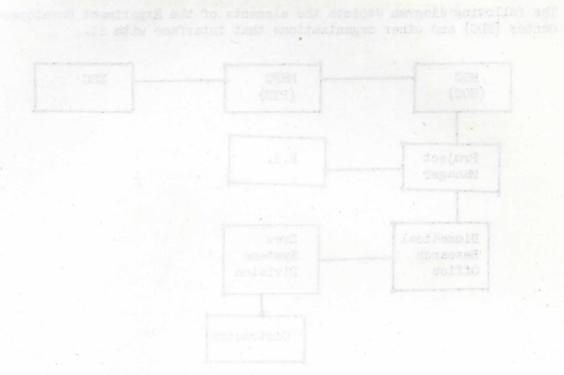
The experiment is to be performed using each of the astronauts (3) as subject for 1.2 hours per occurrence. One astronaut will be required simultaneously as observer. Frequency of experiment is once per day per astronaut at equal intervals throughout the mission. Total manhours: 1.2 hours x 2 men x 3 occurrences per day x $\frac{1}{2}$ days = 28.8 hours.

4.0 PREIAUNCH SUPPORT: Operation and calibration of data sensors and gauges must be checked and recorded.

reporter to the data will be returned in the form of bend to

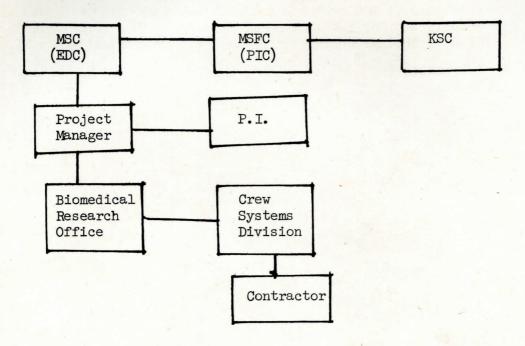
RECOVERY REQUIREMENTS: Experiment equipment will not be recovered. The data will be returned in the form of tape to confirm TM data. Weight = 2.5 lbs., Volume = 8.25 cu. ft.

7.0 DATA SUPPORT REQUIREMENTS: There are no special data support requirements.



SECTION VI - MANAGEMENT ARRANGEMENTS

The following diagram depicts the elements of the Experiment Development Center (EDC) and other organizations that interface with it.



RESPONSIBILITY ASSIGNMENTS: The Principal Investigator/Program Manager will have overall responsibility for integration control of the various elements of the experiments. The Project Managers will assume responsibility for acceptable hardware development and testing. Monitoring of activities will be performed at MSC by the Biomedical Research Office. Preflight studies and the integration of findings into a final report will be the responsibility of the Principal Investigator (PI).

testing; experiment development and commatibility checks will corrier and subsystem mackages; experiments operational procedures development; and experiment logistics comporing plan-

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development will be by the Binmedical Research Calica. The

portion and C. L. Fischer, M.D., for the gra- and pentilight

2.0

ORGANIZATION DESCRIPTIONS: The sponsoring program office is NASA Headquarters, OMSF, with E. McLaughlin, M.D., as the supporting office sponsor. Marshall Space Flight Center is the Experiment Integration Center. The Payload Integration Center will perform all technical and administrative tasks required to assure overall mission coordination between the Experiment Development Center or Principal Investigator and all other NASA organizations. The PIC will provide the resources for experiment receipt and storage, bench testing, installation in carrier, and integrated experiment/carrier testing; experiment development and compatibility checks with carrier and subsystem mockups; experiment operational procedures development; and experiment logistics supporing planning. This group will direct the Kennedy Space Center in installation and spacecraft modifications for the experimental hardware, as well as in retrieval of the tape recordings.

At the Manned Spacecraft Center, supervision of experiment development will be by the Biomedical Research Office. The experiment PI is R. L. Johnson, M.D., for the entire flight portion and C. L. Fischer, M.D., for the pre- and postflight hematology portion.

SECTION VII - MANAGEMENT REPORTING

Financial, schedule, and technical information will be furnished to the appropriate Headquarters program office as required.

Reporting documentation to be used for managing the program are as follows:

- a. Monthly Progress Report
- b. Monthly Financial Reports (inhouse and contractor)
- c. Final Report

SECTION VIII - PROCUREMENT ARRANGEMENTS

All hardware will be developed by MSC. Contractors have not been selected.

SECTION IX

SCHEDULE AND RESOURCES REQUIREMENTS

SECTION IX - SCHEDULE AND RESOURCE REQUIREMENTS 1.0 DEVELOPMENT SCHEDULE

Indicate the planned development schedule for the major milestones outlined below:

			anned Development Schedule	evelop	ment Sc	chedule			
Major Milestones	FY 67	7		FY (89		FY	69 A	
Experiment Approval	7	4					,		
RFP Release		٥					•		
Hardware Contract Awarded	,	٥		Y					
Integration Specification Issued	•								
Interface Control Document. Complete		cocx	4						
Revise EIP		N. S	4						
Design Complete		HON	٥						
Prototype Delivered		PIO		ব		,			
Quality Testing Complete		IR C		<					
Flight Units Fabricated		IWA		1					
Flight Units Assembled		a.ru							
Acceptance Test Complete	*	Gialit		4					
Flight Hardware Delivered		06							
Ground Support Equipment Delivered									
Installation Complete								-	
Checkout Complete									
Shipped to Launch Site									
Flight Analysis Complete									
Final Report									
			-	-	-	-			

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			DESIGN Exp. Hardware Trai ning Equip. Checkout Equip.	SUPPORT STUDIES	SUPPORT FACILITIES AND SERVICES	SUPPORT EQUIPMENT Mockup	Engineering test units	Assembly Test	Prototype	Fabrication Resembly Fer Test	Training Units Fabrication Assembly Test	Checkout Equipment Fabrication Assembly	lest.

44 TOTALS FY QUARTERLY FUNDING REQUIREMENTS (Dollars in Thousands) FY FY Funding Source jo Major Documentation Briefings & Reviews EHICLE INTEGRATION Def. Exp. Plan Rel. Predictions Qual. Test Rept. Failure Analysis FLIGHT EQUIPMENT Periodic Reports DOCUMENTATION Fabrication Vehicle Modif. Fabrication Assembly Testing Assembly Installation Flight Unit Testing Checkout Spares TRAVEL (VI 110) (x1110) xi ko)

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LY FUNDING REQUIREMENTS (Dollars in Thousands)	FY	
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		PUBLICATION Data reduction Analysis Reporting

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3.0 MANPOWER:

3.1 <u>Inhouse (man-months)</u>:

	FY67	FY 68	F Y 69	Total
Contract Technician	_	48	48	96

3.2 <u>Contractor:</u> To be determined during contract negotiations.

4.0 FACILITIES: The following major facilities will be used for conducting the experiment hardware development testing, qualification tests, and astronaut training:

NASA Manned Spacecraft Center (inhouse)

SECTION X - EXPERIMENT RESULTS

DATA HANDLING: Operational biomedical data and voice communications recorded on the ground during this experiment should be provided to Biomedical Research Office, Manned Spacecraft Center. Tapes received from the spacecraft should be returned to the Biomedical Research Office, Manned Spacecraft Center for processing.

2.0	TECHNICAL REPORTS:
2.1	"Quick Look" reports 1 month after flight by PI.
2.2	Data reports 3 months after flight by PI.
2.3	Preliminary analysis 4 months after flight by PI.
2.4	Technical notes/memorandum (NASA and contractor) 8 months after flight by PI.

ATTACHMENT 2

Alterations in erythrocyte and leukocyte kinetics can be expected from intervals of prolonged space flight. Observations already made on Gemini missions show a time related, hemolytic process manifested by a decreased red cell mass, shortened 51CrT1, and an increased spleen/liver ratio. It is, therefore, of considerable interest to monitor the hematologic parameters on any future long duration mission.

Any pre- and postmission program designed to investigate the hematology of manned space flight must take into account the spacecraft atmosphere, crew activity levels, and diet. Therefore, this information should be made available to the personnel interpreting the hematologic results. Furthermore, the formed elements of the blood cannot be separated from the plasma milieu in which they circulate. Thus, a complete chemical and physical profile of plasma must accompany the hematologic effort.

Basically, the hematology effort is designed to investigate the following fundamental aspects of the erythron: red cell production, release, and destruction. Production release information is best obtained by isotope methods, specifically Fe⁵⁹ plasma clearance and reappearance times. Aspects of red cell destruction include quantitation and determination of etiology. The former aspect is best "looked at" by isotope techniques including 51Cr red cell mass determinations and I125 RISA measurements. This combination yields a directly measured total blood The 51CrT1 values are also obtained from the red cell mass volume. measurement, supplying an indication of red cell longevity. The etiologic determination of any hemolytic process is rather detailed. Loss of red cell integrity, with early demise, may be the result of an altered plasma membrane, reduced available energy stores, and/or altered internal osmotic properties. Various specific enzyme systems contribute to various aspects of these vital red cell parameters and, therefore, must be assayed. Various physical measurements of the erythrocyte population; e.g., counts, differentials, osmotic fragilities, etc., are also essential for they supply information needed to integrate the more detailed biochemical analyses. Based on the reasoning used in this text, the appended test protocol is suggested.

ADDENDUM 1

Suggested Hematology Testing

	Responsible	
Test	Agency	Method
Erythrocyte Count	NASA-MSC	Coulter Counter
Het	ii ii	Micro Hematocrit
Hgb	11 11	Cyanomethemaglobin
Osmotic Fragility	11 11 .	Fragiligraph
Reticulocyte Count	11 11	Wet Method
Platelet Count	11 11	Phase Microcopy
WBC	11 11	Routine
Differential	11 11	Routine
Hgb Electrophoresis	11 11	Cellulous Acetate
Plasma Immunoelectrophoresis	11 11	Strips
Erythrocyte Electrolytes	11 11	MgCl Rinse and Analysi
Plasma Osmolality	11 11	Routine
Plasma Electrolytes	11 11	Routine
Trasma Breedier, tes		
Erythrocyte Phosphates Organic	University of Texas	To be stipulated
Inorganic		Allip : Percycles Assay
Red Cell Mass Plasma Volume Iron Kinetics	Baylor University	Cr ⁵¹ in Vitro Tag RI5A - 125 Fe59
Tion minotop		
Red Cell Lipids	University of Toronto	Gas Chromatography
Plasma Lipids	11 11 11	" "
Fecal Urobilingen	11 11 11	" "
Erythrocyte Enzymes	Ohio State University	
and Associated Studies		
Methemoglobin	11 11 11	Spectrophotometry
Autohemolysis	11 11	24 and 48 Hours
Hemolytic Sensitivity to Complement Lysis Acidified Serum	11 11	
Erythrocyte Catalase	11 11 11	Titrametric Method of Feinstein

_bodieN /	Responsible	to the state of th
Test	Agency	Method
Reduced Glutathione	Ohio State University	Spectrophotometric Method for SH Grays
Specific Glycolytic Enzyme Systems Glucose 6-PO ₄ Pyrnvate Kinase Phosphoglyceraldehyde Dehydrogenase Phosphofructokinase Red Gell Cholinesterase	H H H	Acetylthicholine Spectrophotometric Method
	Same To vilenested	Erythmoyte Phosphales
H202 Lysis Test "Lipid Peroxids" Assay in vivo and in vitro	11 11 11	Method of Mengel
	Baylor University	need (for bed
Glutathione Peroxidase	п п п п	H ₂ O ₂ Diffusion Tech- nique
Spectrophotosetry 24 and 48 Hours		end Associated Studies ,Methemoglobin Autohemolysis Remolytic Schaltsvity to Complement Lysis
		Acidified Serum Erychronyte Catalase

																							1		
	Radiation Dose	88 m rem											268 m rem										88 m rem	444 m rem	
	5 3	50	10	10		17				17			. 5	CU	2	CI	25	10	10	25	7	7	50	249	
	Blood Requirements Number of Venopunctures	3				1				1			3	1	1	1	1	1	1	1	1	1	3	20	
II.	Special Hematology	X		X		X				X			X				X	×	×	X	X	X	X		
ADDENDUM II	Fe ⁵⁹ Kinetics												X	X	X	X	X	X	X	×				SILS	
	Red Cell Survival		X	X		X				X							X	X	X	X	X	X	X	TOTALS	
	Plasma Volume Test	X											X									- 1	X		
	Red Cell Mass	X											X					,					X		
	Plasma Membrane Assay	X				X				X			X				X			X			X		
	Enzyme Assay	X				X	1			X			X				X			X			X		
	Day	F-10	F-9	F-8	F-7	F-6	F-5	1−4	F-3	F-2	F-1	ഥ	R(ASAP)	R+30 min	R+60 min	R+90 min	R+1 day	R+3 days	R+5 days	R+7 days	R+14 days	R+21 days	R+30 days		

MSC FORM 753 (FEB 63)